

Planetary Laser Altimetry; Past and Present

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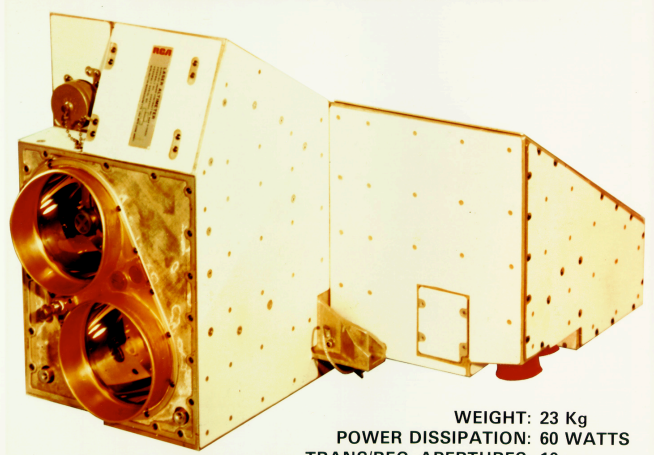
16th International Workshop on Laser Ranging,
Poznan, Poland

October 13-17, 2008

After Apollo, laser altimetry in NASA began with Mars Observer (MOLA-1), launched in 1992 but lost on approach to Mars, and re-flown on Mars global Surveyor (MOLA-2) in 1996. In Jan 1994 the Clementine mission to the Moon carried the second laser altimeter into space that provided the first global shape of the Moon. In 1996 the NEAR spacecraft carrying the NEAR Laser Ranger (NLR) was launched to the asteroid 433 Eros, and in 2004 a laser altimeter was launched on the MESSENGER spacecraft to Mercury and is currently in cruise to the planet. Early next year the LRO spacecraft carrying LOLA will be launched to the Moon. Japan launched the SELENE spacecraft in 2007 carrying the LALT laser altimeter followed by Chang'E by China. Soon India will launch Chandrayaan carrying a laser altimeter. In nearly all the missions to date laser altimeters have been used for mapping of planetary bodies with remarkable success and played major roles in preparing for subsequent lander missions. All of these missions and their laser instruments have helped make advances in the use of lasers for planetary science and helped convince skeptical space agencies that these kinds of instruments could be used with confidence and reliability on long planetary missions in some harsh environments. Laser altimetry is now accepted, albeit not with the same level of confidence as microwave instruments, and laser tracking of planetary spacecraft will be next challenge.

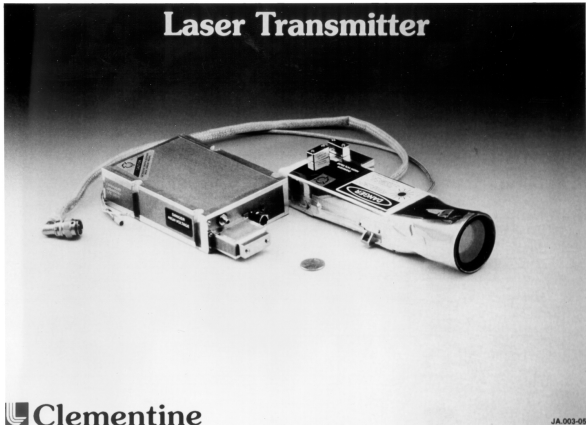
US Planetary Laser Altimeters 1969 - 2009

APOLLO LUNAR LASER ALTIMETER



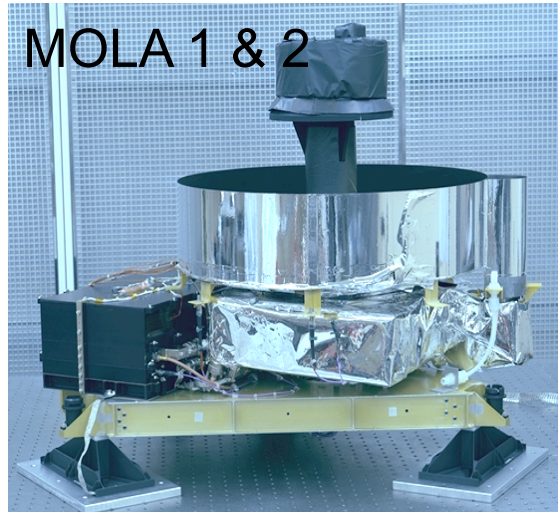
WEIGHT: 23 Kg
POWER DISSIPATION: 60 WATTS
TRANS/REC. APERTURES: 10 cm

Laser Transmitter

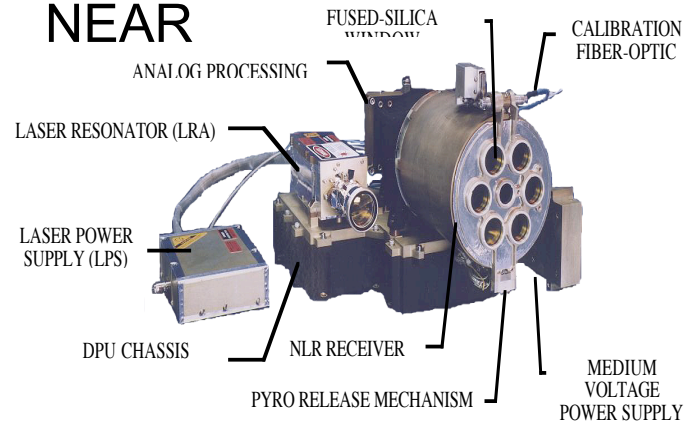


Clementine

MOLA 1 & 2



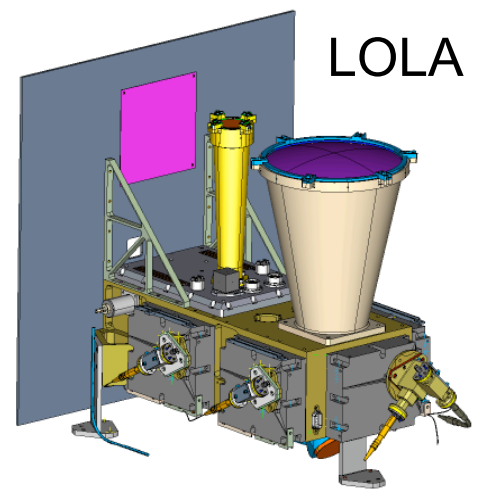
NEAR



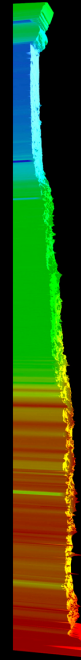
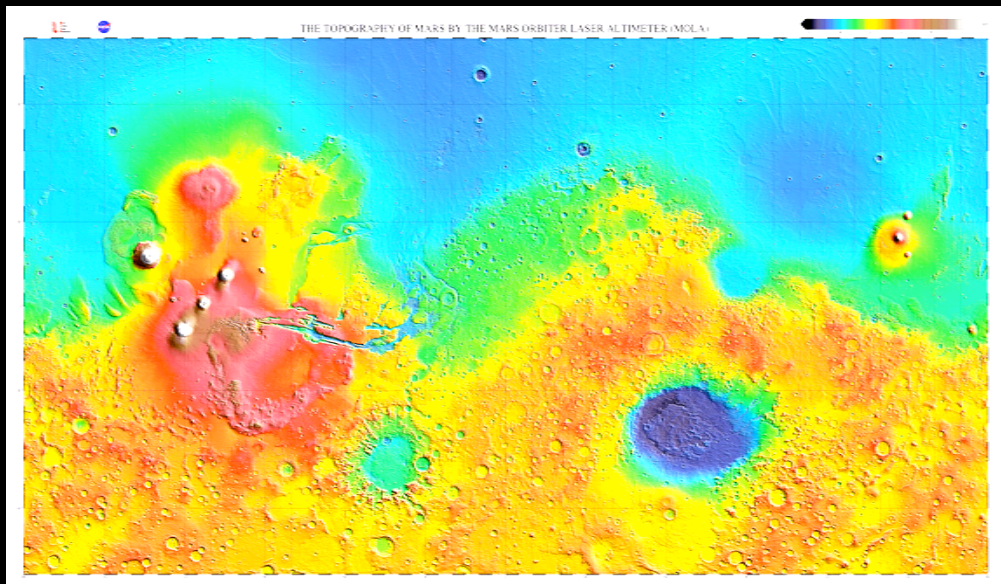
MESSENGER



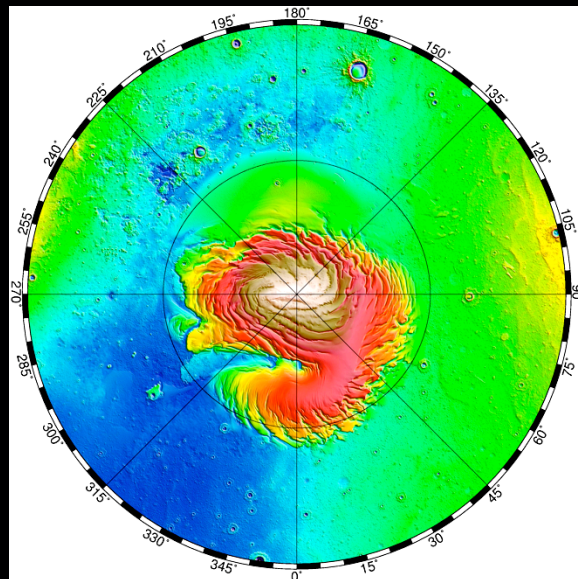
LOLA



Global Topography of Mars from MOLA

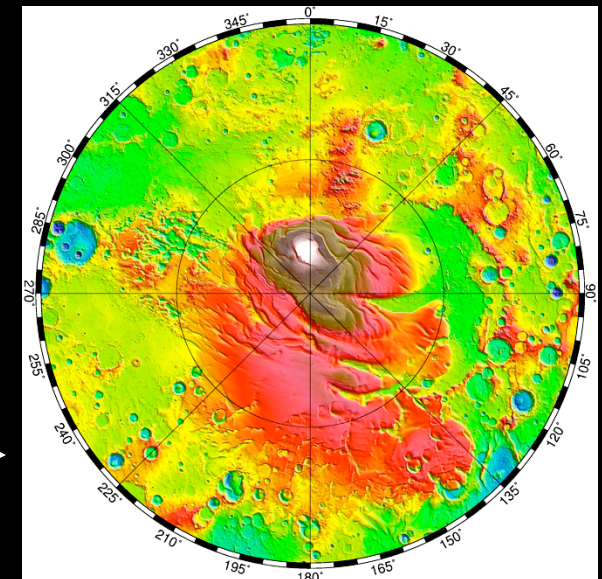


Equatorial topography and South to North slope



← North Pole

South Pole →

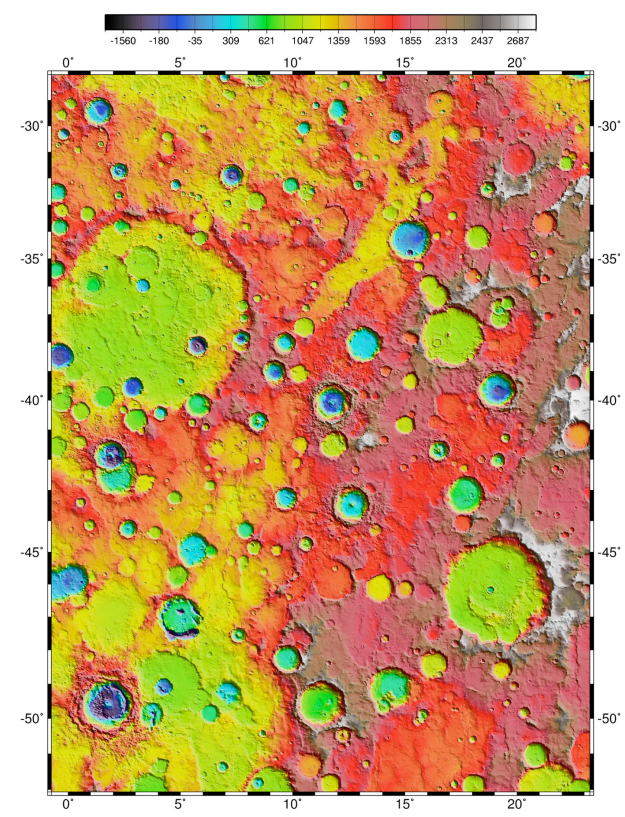
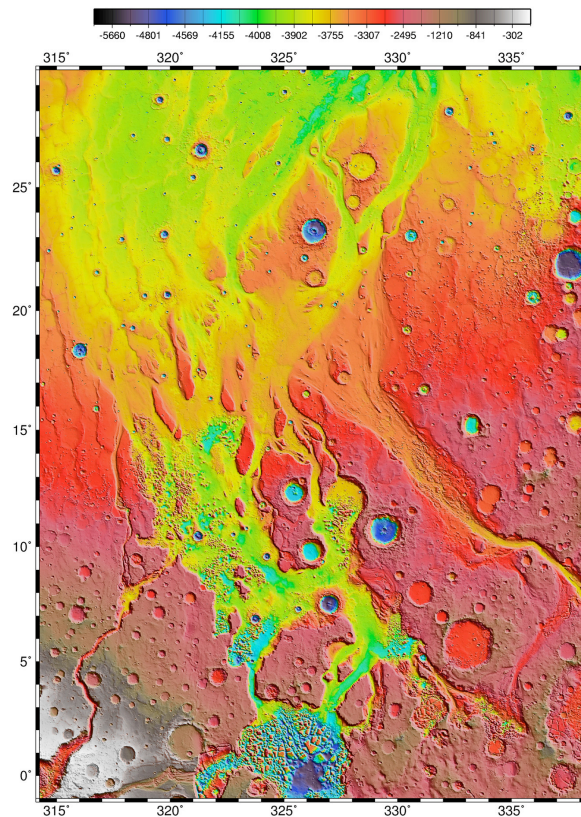
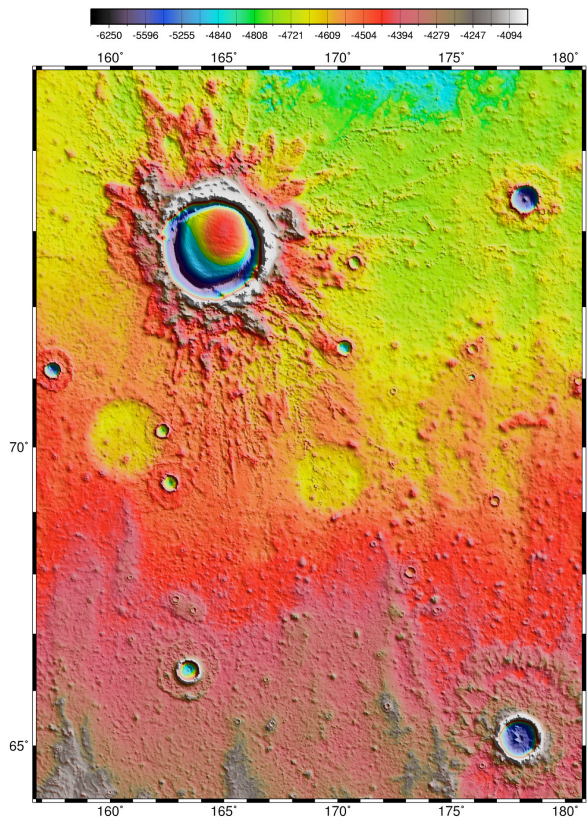


Topographic Images from Altimetry

Ice filled crater

Outflow channels

Cratered highlands

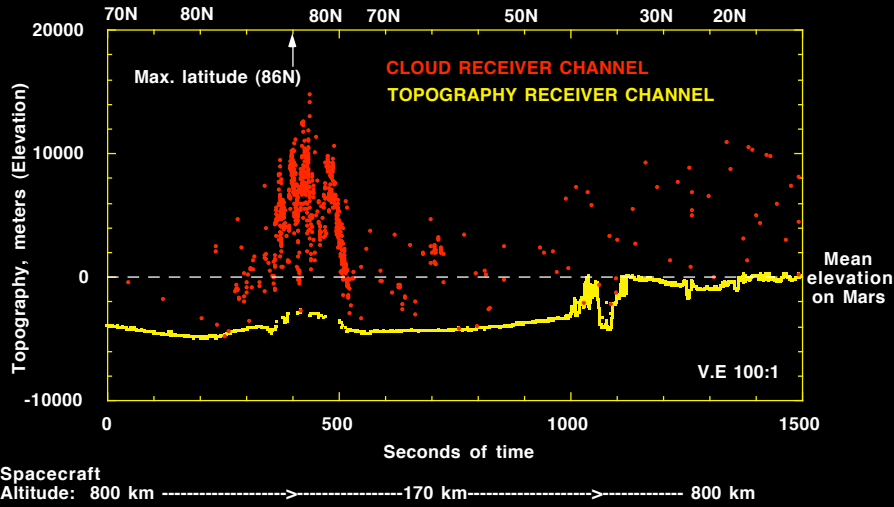


660 x 430 km

1800 x 1400 km

1800 x 1000 km

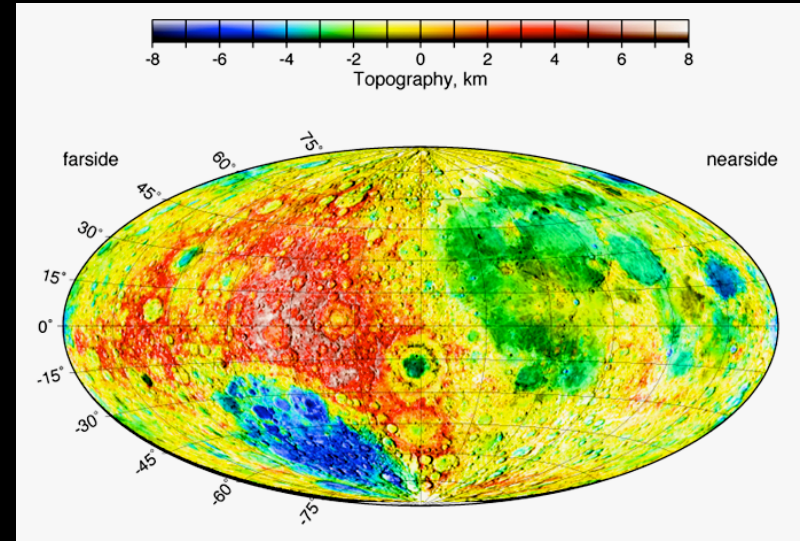
Clouds over Mars' North Pole



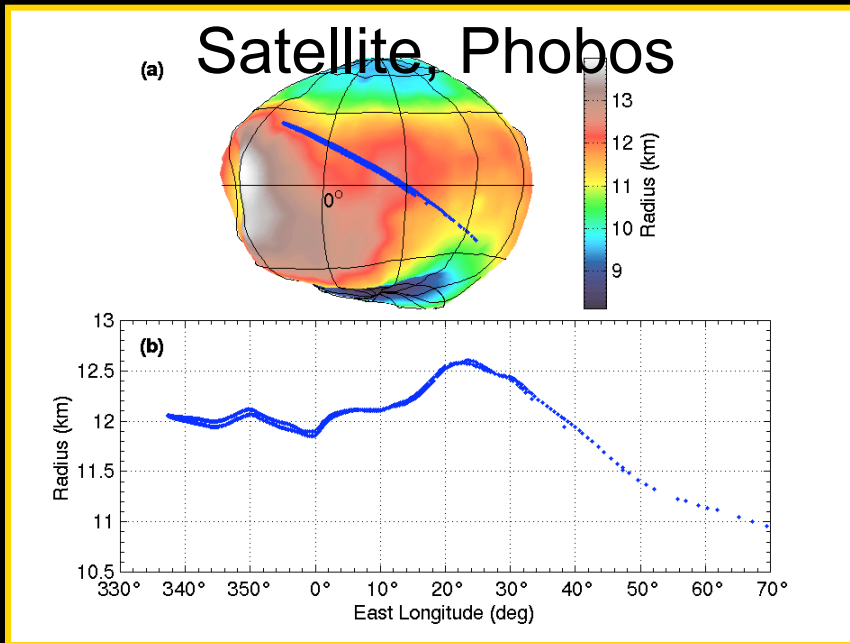
MGS Pass P226a on 4/7/98

MOLA Science Team

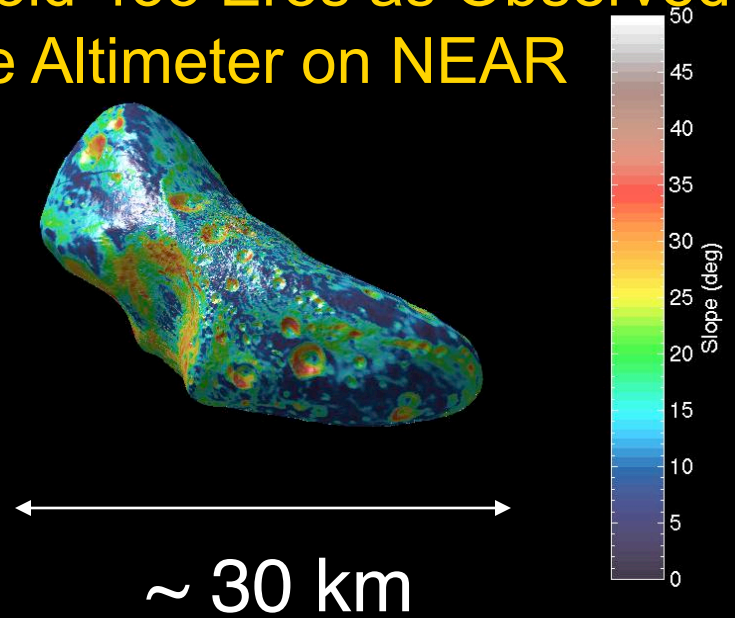
Lunar Topography by Clementine



MOLA Profile Across Mars'



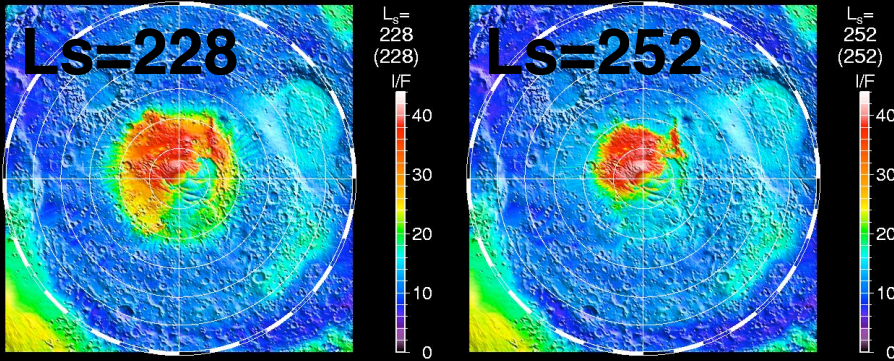
Asteroid 433 Eros as Observed by the Altimeter on NEAR



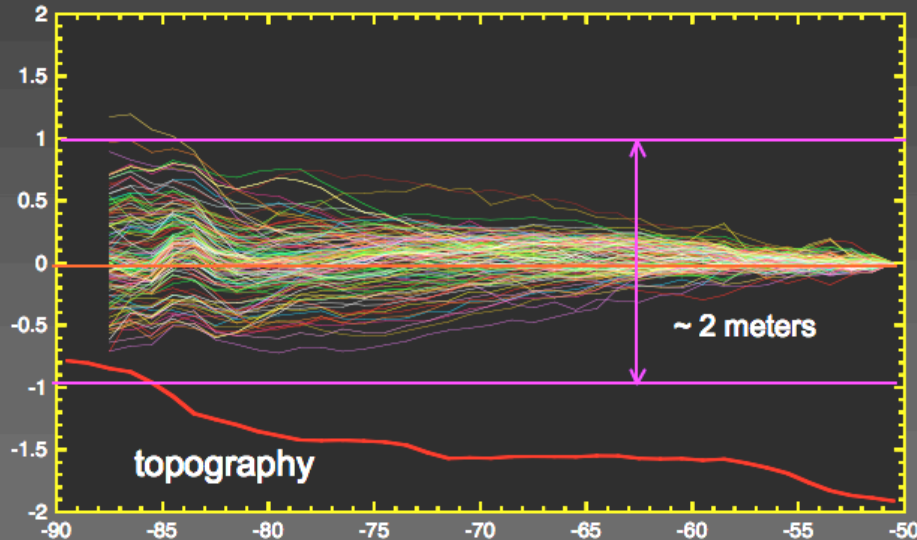
Monitoring the Seasonal Changes in Mars Ice Caps

South Polar Seasonal Icecap

Radiometry

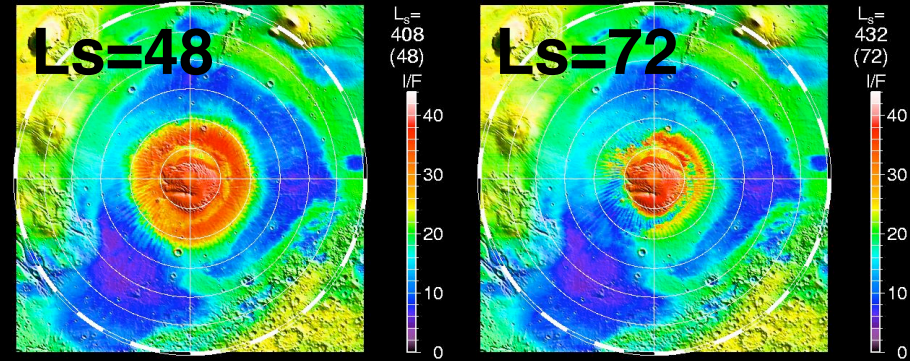


Elevation

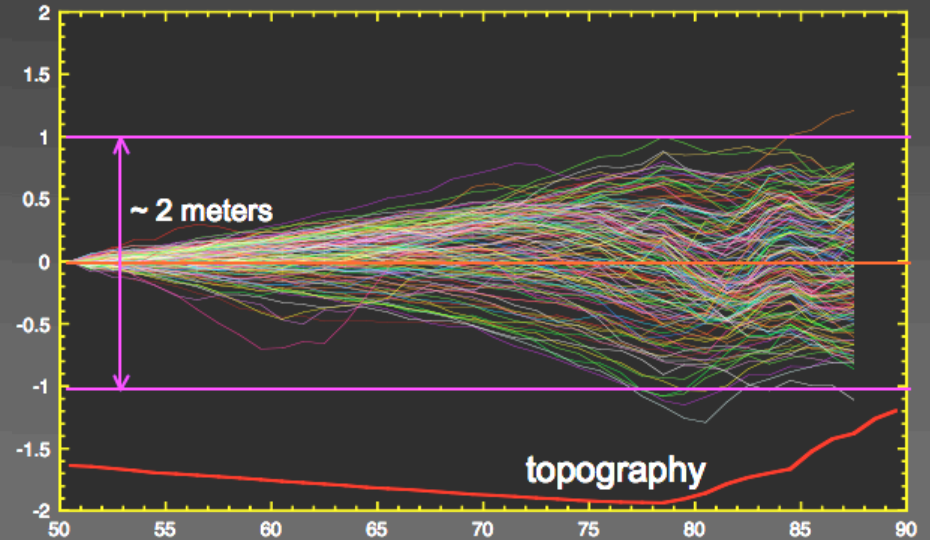


North Polar Seasonal Icecap

Radiometry



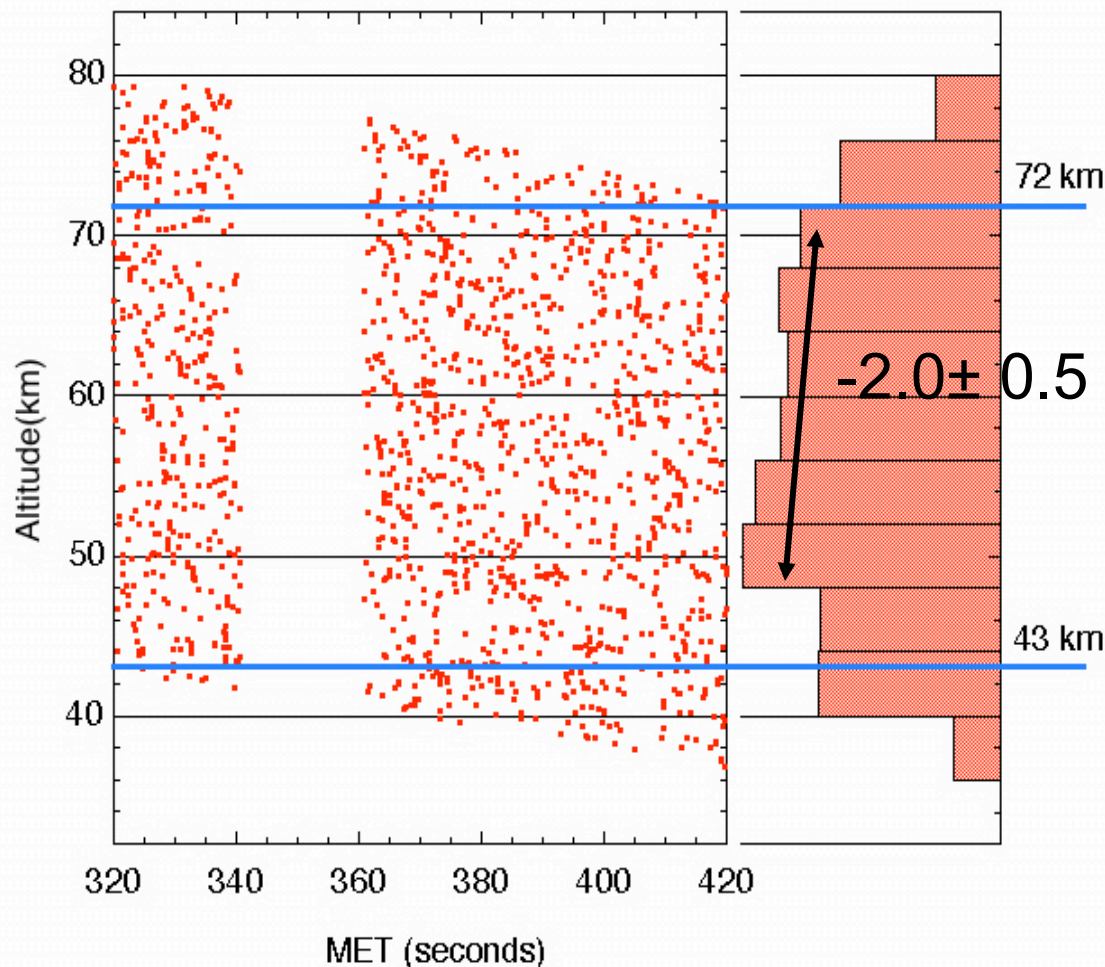
Elevation



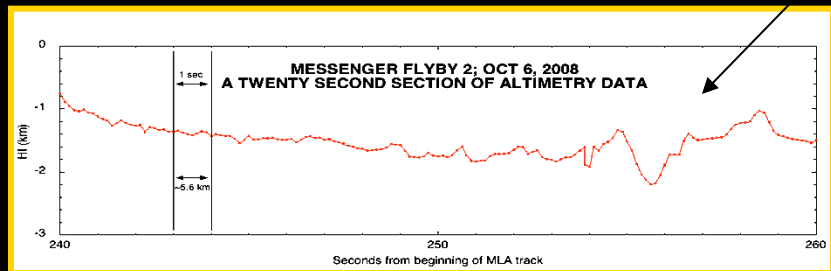
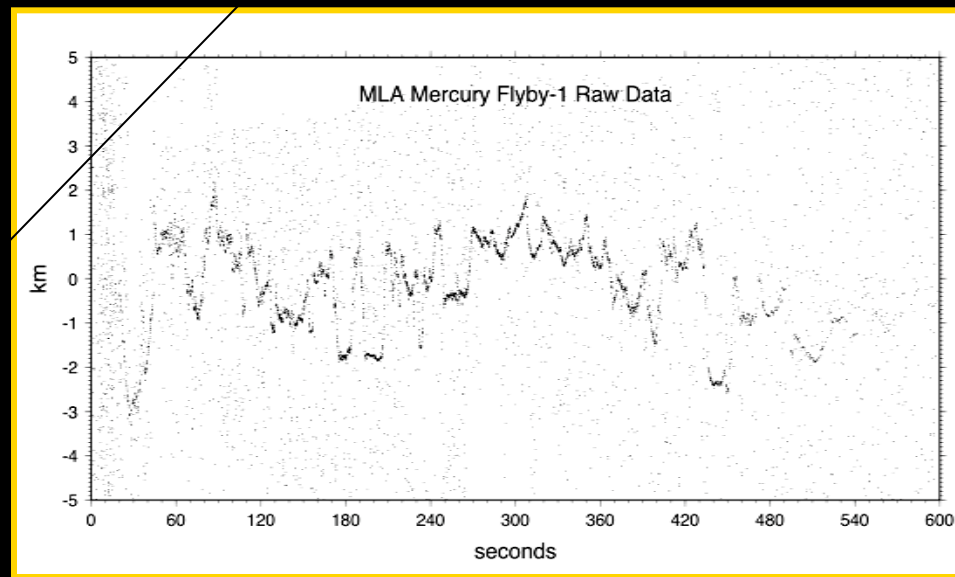
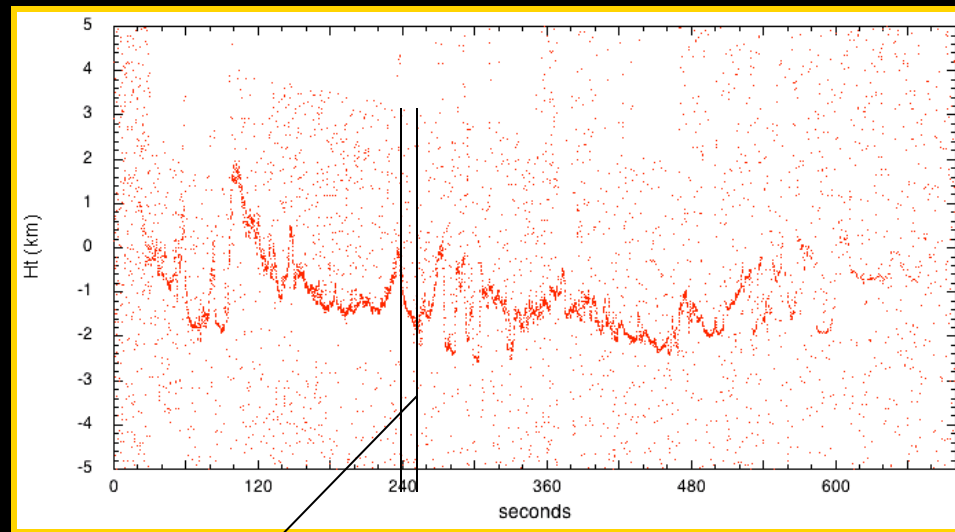
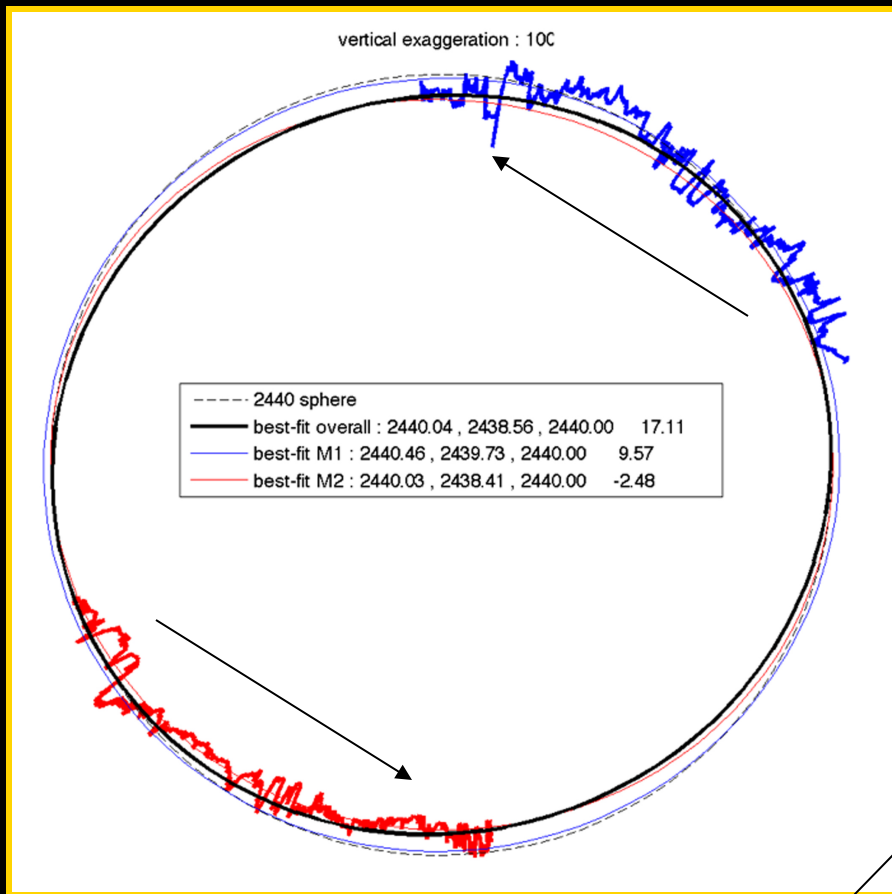
MESSENGER Passes by Venus, Altimeter Ranges to the Atmosphere

- Detector counts are shown as a function of altitude and MET.
- Number of counts in each 4-km-wide bin is shown on right.
- Number of counts increases slightly but steadily as altitude decreases, reaching a maximum at ~50 km.
- Outside blue lines the number of counts decreases due to the lack of data at all MET's.

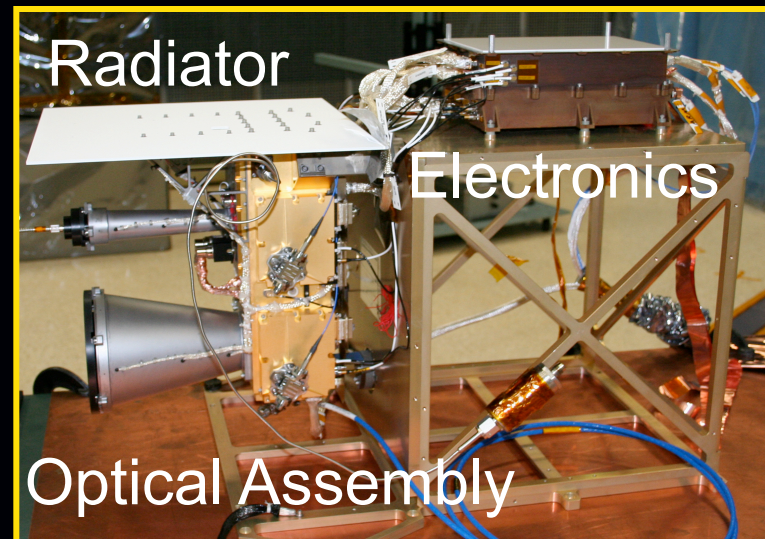
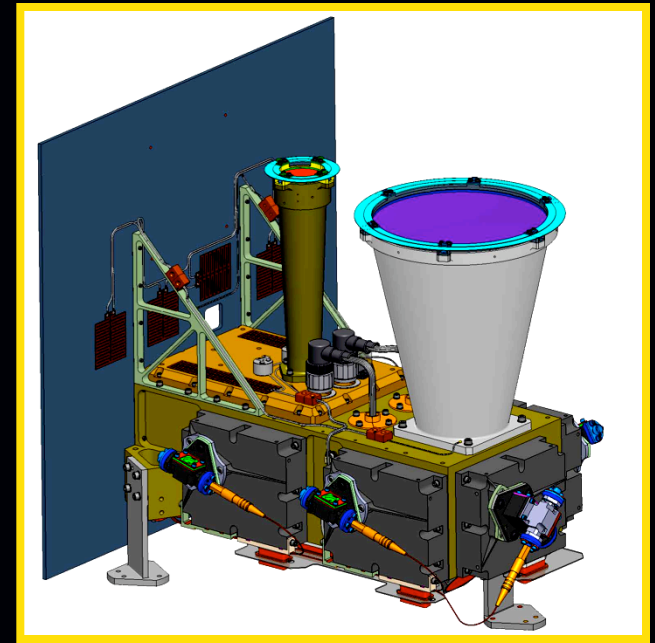
Channel 3 optimized for 270 ns (40 m)



MESSENGER Flies By Mercury, Jan and Oct, 2008

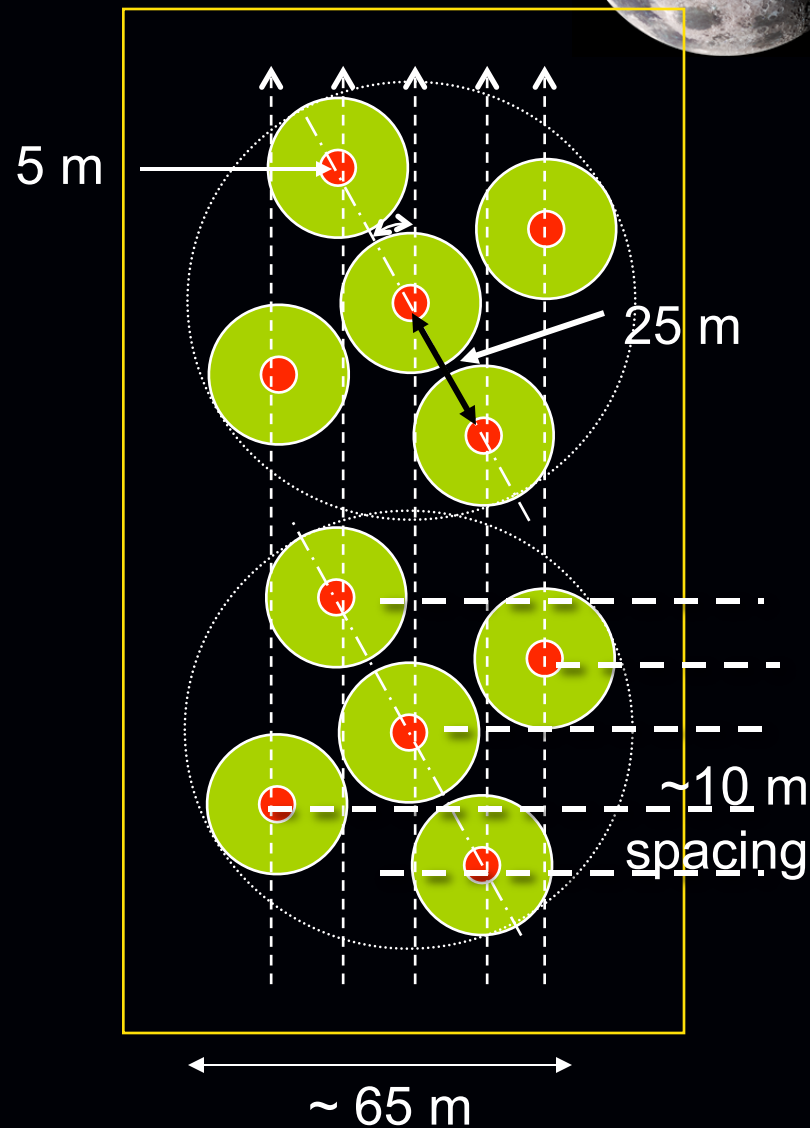


- LOLA will characterize a swath ~ 50 meters wide under LRO ground-track.
- LOLA will make 140 measurements/second of range to surface below spacecraft.
- Every observation will be registered to a global lunar center-of-mass coordinate system.
- LOLA designed to provide data necessary to identify locations for safe landing of robotic spacecraft.





- In each 5 meter spot:
 - Range to surface
 - Roughness of surface
 - Reflectance of surface
- LOLA operates continuously
- Transmitter Energy: 2.7 mJ
- DOE: 5 Beams, 100 μ rad div.
- Receiver: 0.14-m diameter
- FOV: 400 μ rad
- Detectors: 5 fiber optically-coupled avalanche photodiodes



Surface Slopes

- Within a 100 m x 50 m along-track area there will be 10 5-m spots.
- There will be 22 slopes with baselines ranging from 25 m to 50 m.
- There are another 23 slopes with baselines of >50 to <100 m.
- For slopes $< \pm 3^\circ$ in a 5-m spot, height precision will be $< \pm 10$ cm and slope accuracies will be $\pm 0.3^\circ$ to 0.15° .

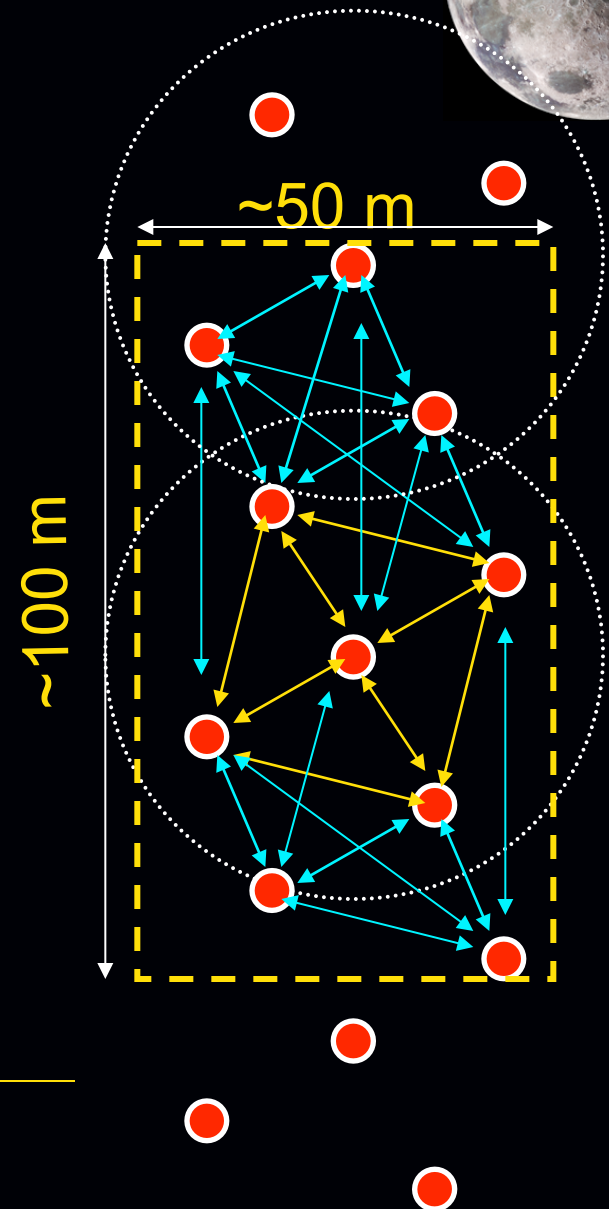
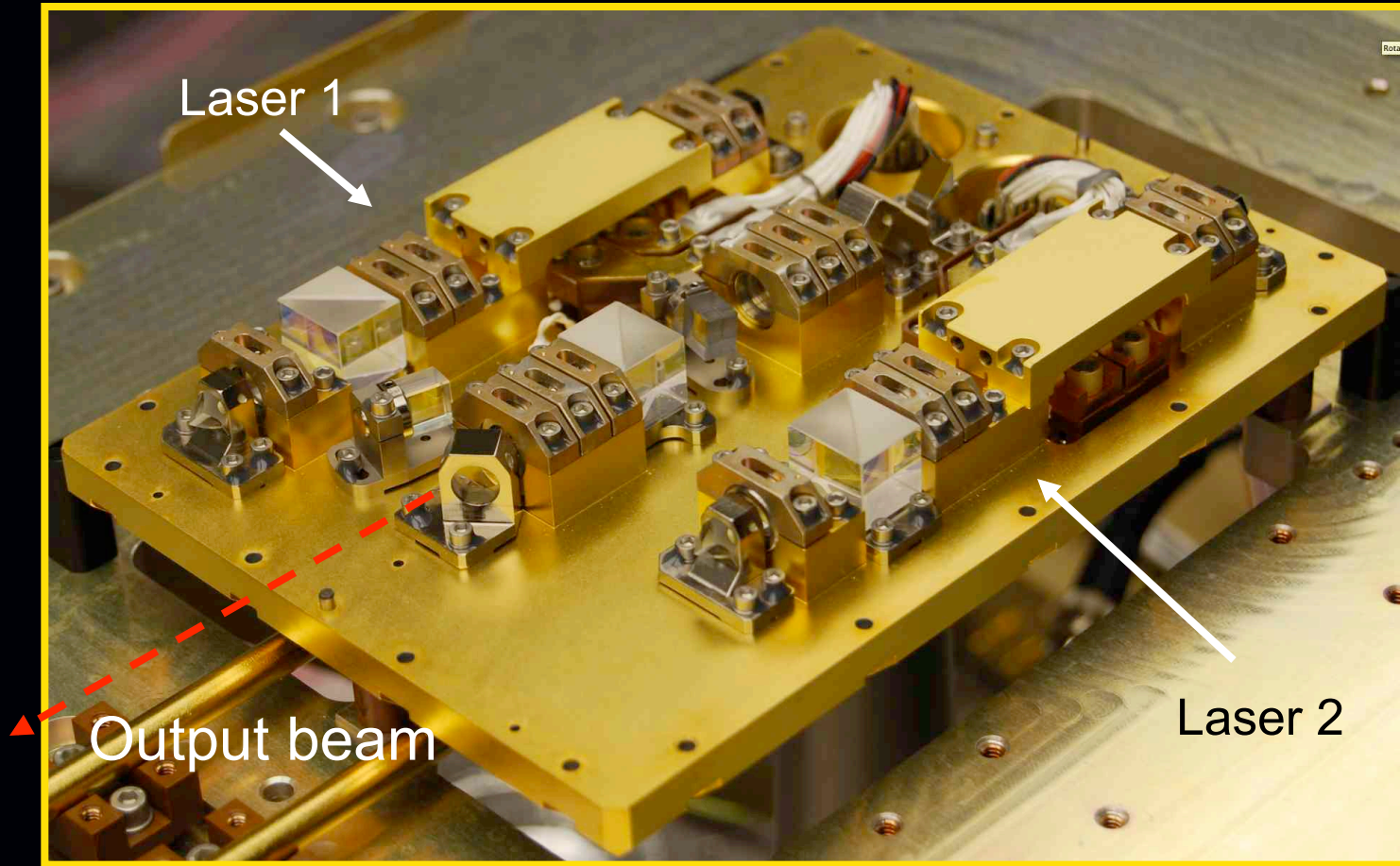


Chart shows 22 measured baselines <50 m

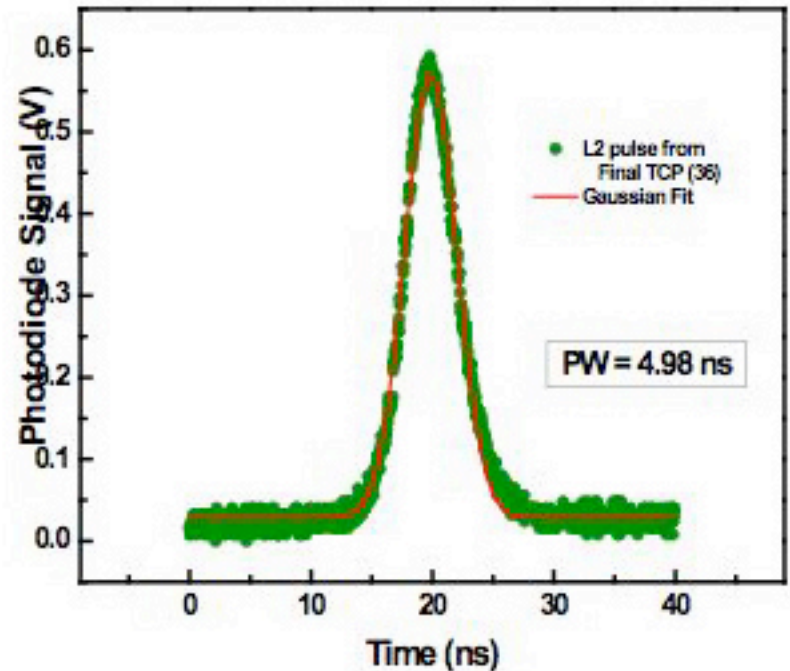
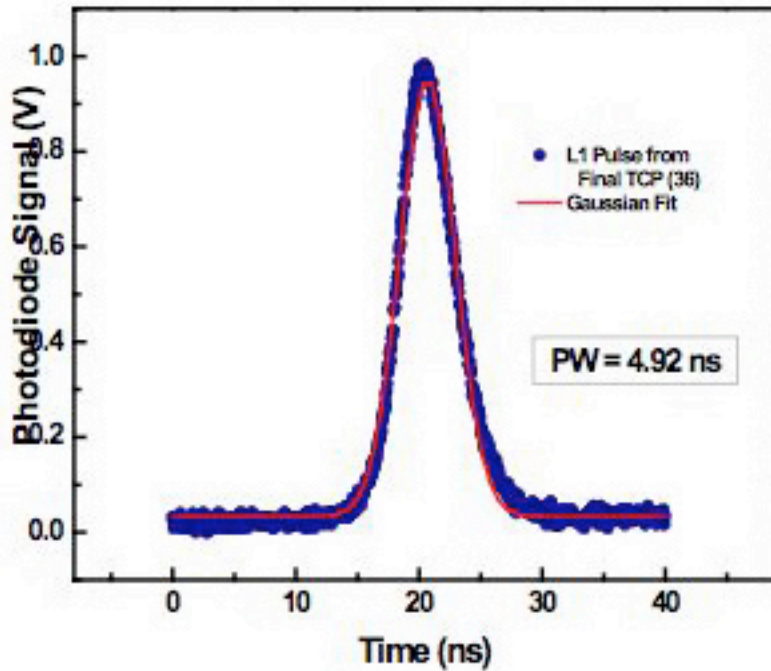
Flight Model Laser



Flight lasers on flight laser bench

Laser Pulse Width

Outgoing pulse width is < 5 ns for lasers 1 and 2

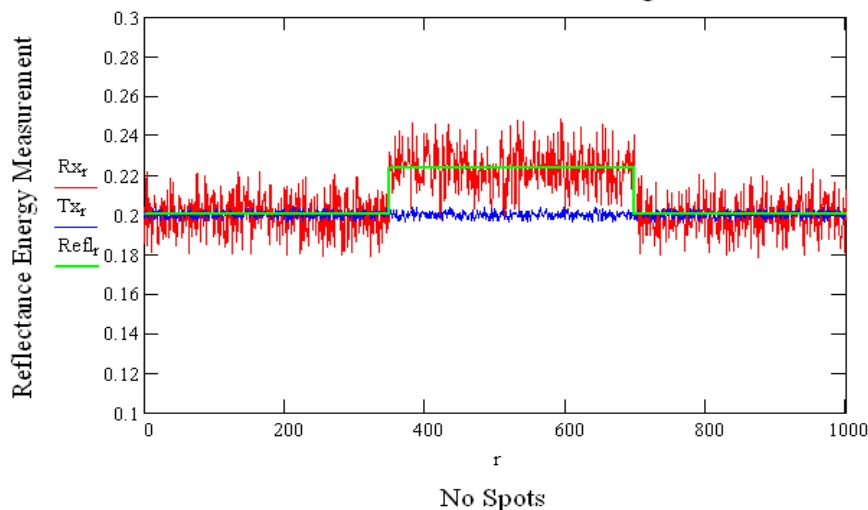


Timing resolution 30 ps; accuracy 100 ps (3 cm)

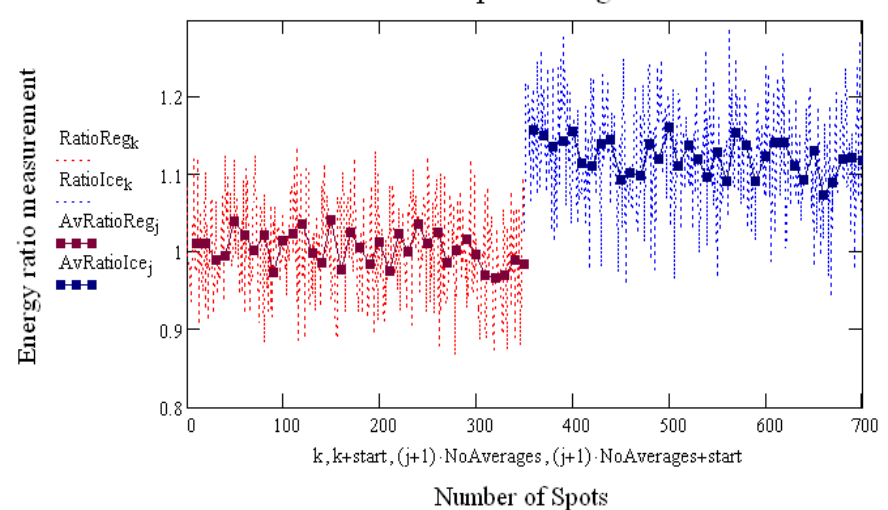
Surface Reflectance Measurement

- LOLA will measure surface reflectance to laser light via ratio of transmitted (Tx) and echo pulse energies (Rx) to search for ice in permanently shadowed regions.
- Mixtures of ice crystals in regolith (4% ice) will exhibit a higher surface reflectance ($\beta = 80\%$ ice, $\beta = 20\%$ regolith) than surrounding areas.

Reflectance of 4% Ice in Regolith



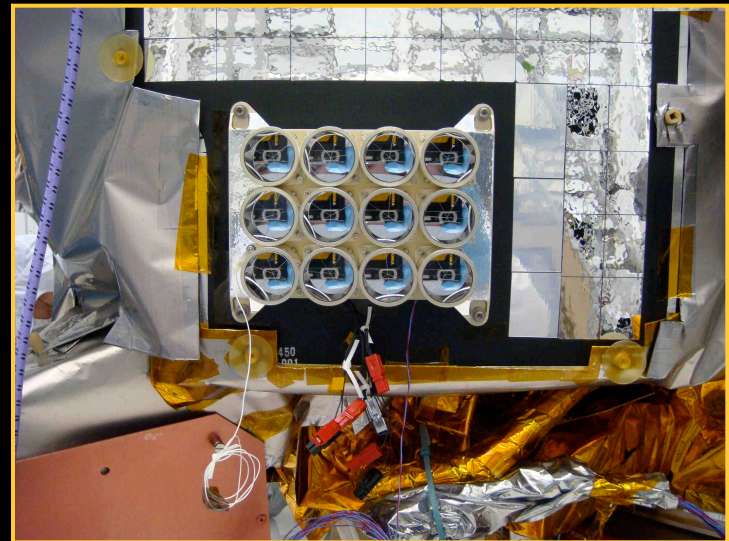
10 Spot Average



Simulated Regolith-Ice mixture detection assuming 2% noise in Tx and 12% in Rx

Laser Corner Cube Array (LArray)

- 12, 31.7mm diameter solid corner cubes
- Materials:
 - Suprasil (quartz) cubes, Al frame
- Mass: 650.25 g
- Volume: 15x18x5 cm
- Oper. temp range: -150 to +30 C
Thermally isolated from the spacecraft
- Tested to 14g vibration.
- Optical characteristics:
 - 90 deg. dihedral angle (unspoiled),
with 0.3 arcsec tolerance
 - Total internal reflection
(uncoated back surfaces)
 - AR coating on top surface





Altimetry as Art; Mars by MOLA and Kees Veenebos

LOLA

Lunar Orbiter Laser Altimeter



~ 100km

Shaded relief of Mars' south polar layered terrains from MOLA
A view a camera can never take.